

Homework 8

Due Date: Mon, Dec 2nd, 2019 at 11:59pm

1. Lyapunov Equation and Stability

- (a) **(PTS: 0-2)** Show that if A is stable, then $P = \int_0^\infty e^{A^T t} Q e^{At} dt$ solves

$$A^T P + PA + Q = 0$$

- (b) **(PTS: 0-2)** Show that if there exists $P = P^T > 0$ that solves

$$A^T P + PA + Q = 0$$

for $Q = Q^T > 0$, then A is stable and find a bound on the largest real part of the eigenvalues of A in terms of the eigenvalues of Q and P .

Suppose $V(x)$ is positive definite, ie. $V(x) > 0 \forall x \neq 0$, $V(0) = 0$, $V(x)$ has bounded level sets.

- (c) **(PTS: 0-2)** Show that if $\frac{\partial V}{\partial x} f(x) < 0$, then $x = 0$ is asymptotically stable.
 (d) **(PTS: 0-2)** Show that if $\frac{\partial V}{\partial x} f(x) < \mu V(x)$, then $x = 0$ is exponentially stable. What is an upper bound on the decay rate?

2. Hamiltonian Systems for Riccati Equations

Let

$$A = \begin{bmatrix} -2 & 2 & -1 \\ 3 & 1 & 0 \\ -1 & 0 & 1 \end{bmatrix}, \quad B = \begin{bmatrix} 1 \\ 1 \\ 0 \end{bmatrix}$$

$$Q = \begin{bmatrix} 1 & 0 & 0 \\ 0 & 1 & 0 \\ 0 & 0 & 1 \end{bmatrix}, \quad R = [1], \quad Q_T = \begin{bmatrix} 1 & 0 & 0 \\ 0 & 1 & 0 \\ 0 & 0 & 1 \end{bmatrix}$$

- (a) **(PTS: 0-2)** Use the Hamiltonian system to compute the solution $P(t)$ to the Riccati differential equation

$$-\dot{P} = A^T P + PA + Q - PBR^{-1}B^T P, \quad P(T) = Q_T$$

Show the calculation method and then write down $P(t)$ at several different time steps.

- (b) **(PTS: 0-2)** Use the Hamiltonian matrix to compute the solution to the algebraic Riccati equation

$$0 = A^T P + PA + Q - PBR^{-1}B^T P$$

3. Adjoint Method for Optimal Control

Consider the problem of finding the shape of a slope that causes a frictionless stone to slide the farthest in a given period of time.

This problem can be formulated as

$$\begin{aligned} & \max_{\substack{\theta(t) \text{ for} \\ t \in [0, T]}} J = x(T) \\ \text{s.t.} \quad & \underbrace{\begin{bmatrix} \dot{x} \\ \dot{y} \\ \dot{v} \end{bmatrix}}_z = \underbrace{\begin{bmatrix} v \cos(\theta) \\ v \sin(\theta) \\ -g \sin(\theta) \end{bmatrix}}_{f(z, \theta)}, \quad \begin{bmatrix} x(0) \\ y(0) \\ v(0) \end{bmatrix} = \begin{bmatrix} 0 \\ 0 \\ 0 \end{bmatrix} \end{aligned}$$

where x and y are the horizontal and vertical positions respectively, v is the velocity, $g = 9.81 \text{ m/s}^2$ is the acceleration due to gravity, the cost $J = x(T)$ is the final horizontal position, and $\theta(t)$ is the angle of the slope (from horizontal) at time t . Let $z \in \mathbb{R}^3$ be the full state vector $z = [x \ y \ v]^T$.

- (PTS: 0-2)** Write expressions for $\frac{\partial f}{\partial z}$ and $\frac{\partial f}{\partial \theta}$.
- (PTS: 0-2)** Write out the backwards differential equation for computing the co-state $p(t)$ from the terminal condition $p(T) = \frac{\partial J}{\partial z(T)} = [1 \ 0 \ 0]$ and the expression for computing $\frac{\partial J}{\partial \theta(t)}$ from the co-state $p(t)$.
- (PTS: 0-2)** Write code to compute an ascent direction $\frac{\partial J}{\partial \theta(t)}$ for the cost J with respect to the slope at each time $\theta(t)$.
- (PTS: 0-2)** Initializing $\theta(t) = 0$ for $t \in [0, T]$ (a flat slope), write code to update the slope using gradient ascent with step size α according to the equation

$$\theta^+(t) = \theta(t) + \alpha \frac{\partial J}{\partial \theta(t)}$$

until the cost converges. You can work in discrete time or continuous time. If you use discrete time, you can assume the simple discretization scheme

$$z[t+1] = z[t] + \Delta t f(z, \theta)$$

Use $\alpha = 0.01$, $\Delta t = 0.1$, and $T = 20$ seconds (200 time steps).

- (PTS: 0-2)** Plot the resulting optimal slope as well as several of the suboptimal slopes from the earlier iterations.

Note Problems 4 and 5 will be discussed in class on **Tuesday (Nov 26)**. They deal with the H_2 and H_∞ design methods.

4. Optimal Controller Synthesis (CHOOSE 4 OR 5)

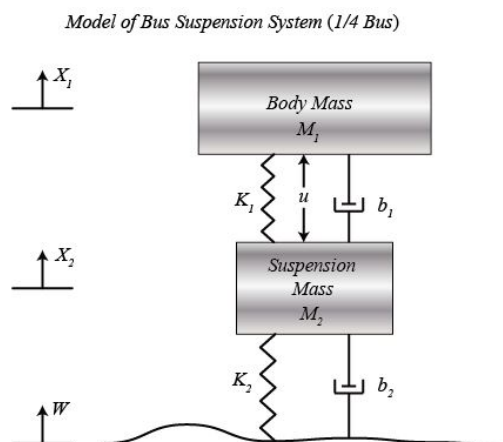
Consider the simple harmonic oscillator (spring-mass-damper) as a generalized process G ,

$$\begin{aligned} m\ddot{x} &= -kx - b\dot{x} + u + \mu \\ y &= x + \eta \\ \omega &= (\mu, \eta), \quad z = (x, u) \\ m = k = b &= 1 \end{aligned}$$

- Compute the H_2 and H_∞ norm of the open-loop transfer function $G_{z\omega}$, set $u = 0$, and plot step response.
- Plot the open-loop sensitivity transfer function $1/(1 + G_{yu})$ and compute the gain, phase, and stability margins.
- Synthesize an H_2 - optimal state-feedback controller K_2 treating ω and z as an error to be minimized with respect to the standard \mathcal{L}_2 norm, compute, the H_2 norm of the closed-loop transfer function and compute the gain, phase and stability margin.
- Same as Part (c) but use the H_∞ norm.
- Discuss Part (a) to Part (d), do the result make sense? What are the difference between how H_2 and H_∞ have the sensitivity transfer function and margins? Which controller would you choose? When and why?
- Same as Part (c) but synthesize an output-feedback controller.
- Same as Part (d) but synthesize an output-feedback controller.

5. Robust Control and Disturbance Rejection

Consider the suspension system.



System Parameters

- $M_1 = 1/4$ bus body mass 2500 [kg]

- (b) $M_2 =$ suspension mass 320 [kg]
- (c) $K_1 =$ spring constant of suspension system 80,000 [N/m]
- (d) $K_2 =$ spring constant of wheel and tire 500,000 [N/m]
- (e) $b_1 =$ damping constant of suspension system 350 [N · s/m]
- (f) $b_2 =$ damping constant of wheel and tire 15,020 [N · s/m]
- (g) $U =$ control force
- (h) $W =$ road disturbance

Equations of Motion:

$$M_1 \ddot{X}_1 = -b_1 (\dot{X}_1 - \dot{X}_2) - K_1 (X_1 - X_2) + U$$

$$M_2 \ddot{X}_2 = b_1 (\dot{X}_1 - \dot{X}_2) + K_1 (X_1 - X_2) + b_2 (\dot{W} - \dot{X}_2) + K_2 (W - X_2) - U$$

State-space:

$$\begin{bmatrix} \dot{X}_1 \\ \ddot{X}_1 \\ \dot{X}_2 \\ \ddot{X}_2 \end{bmatrix} = \begin{bmatrix} 0 & 1 & 0 & 0 \\ \frac{-b_1 b_2}{M_1 M_2} & 0 & \left[\frac{b_1}{M_1} \left(\frac{b_1}{M_1} + \frac{b_1}{M_2} + \frac{b_2}{M_2} \right) - \frac{K_1}{M_1} \right] & \frac{-b_1}{M_1} \\ \frac{b_2}{M_2} & 0 & - \left(\frac{b_1}{M_1} + \frac{b_1}{M_2} + \frac{b_2}{M_2} \right) & 1 \\ \frac{K_2}{M_2} & 0 & - \left(\frac{K_1}{M_1} + \frac{K_1}{M_2} + \frac{K_2}{M_2} \right) & 0 \end{bmatrix} \begin{bmatrix} X_1 \\ \dot{X}_1 \\ X_2 \\ \dot{X}_2 \end{bmatrix} + \begin{bmatrix} 0 \\ \frac{1}{M_1} \\ 0 \\ \left(\frac{1}{M_1} + \frac{1}{M_2} \right) \end{bmatrix} \begin{bmatrix} U \\ W \end{bmatrix}$$

$$Y = \begin{bmatrix} 0 & 0 & 1 & 0 \end{bmatrix} \begin{bmatrix} X_1 \\ \dot{X}_1 \\ X_2 \\ \dot{X}_2 \end{bmatrix} + \begin{bmatrix} 0 & 0 \end{bmatrix} \begin{bmatrix} U \\ W \end{bmatrix}$$

$$\begin{bmatrix} \dot{x}_1 \\ \dot{x}_2 \\ \dot{x}_3 \\ \dot{x}_4 \end{bmatrix} = \begin{bmatrix} 0 & 1 & 0 & 0 \\ -6.571 & 0 & -25.26 & -0.14 \\ 46.94 & 0 & -48.17 & 1 \\ 1563 & 0 & -1845 & 0 \end{bmatrix} \begin{bmatrix} x_1 \\ x_2 \\ x_3 \\ x_4 \end{bmatrix} + \begin{bmatrix} 0 & 0 \\ 0.0004 & 6.571 \\ 0 & -46.94 \\ 0.003525 & -1563 \end{bmatrix} \begin{bmatrix} U \\ W \end{bmatrix}$$

$$Y = \begin{bmatrix} 0 & 0 & 1 & 0 \end{bmatrix} \begin{bmatrix} x_1 \\ x_2 \\ x_3 \\ x_4 \end{bmatrix} + \begin{bmatrix} 0 & 0 \end{bmatrix} \begin{bmatrix} U \\ W \end{bmatrix}$$

- (a) **(PTS: 0-2)** Use state feedback setup to compute the continuous time infinite-horizon LQR controller gain K with Q and R specified below. What is the gain matrix and the norm of

the close-loop transfer function?

$$Q = \begin{bmatrix} 10^3 & 0 & 0 & 0 \\ 0 & 10^5 & 0 & 0 \\ 0 & 0 & 3 \cdot 10^4 & 0 \\ 0 & 0 & 0 & 10^6 \end{bmatrix}, \quad R = \begin{bmatrix} 1 \end{bmatrix}$$

- (b) **(PTS: 0-2)** Use state feedback setup to compute the \mathcal{H}_2 and \mathcal{H}_∞ controller replicating the result of LQR controller in Part (a). What are the gain matrix and the norm of the close-loop transfer function?
- (c) **(PTS: 0-2)** Use state feedback setup to compute the \mathcal{H}_2 and \mathcal{H}_∞ controllers minimize x_3 and u with weighting $W = 10^5$ on x_3 . What are the gain matrix and the norm of the close-loop transfer function?
- (d) **(PTS: 0-2)** Compare the transfer function response (Bode) of LQR, \mathcal{H}_2 , and \mathcal{H}_∞ from disturbance to x_3 . 5 Bode plots.
- (e) **(PTS: 0-2)** Simulate the closed loop system of LQR, \mathcal{H}_2 , and \mathcal{H}_∞ for 10s with a 0.1-m step disturbance input starting at 1s with zero initial condition and plot x_3 . 5 simulations.
- (f) **(PTS: 0-2)** Use output feedback setup to compute the continuous time infinite-horizon LQR controller gain K with Q and R specified in Part (a). What is the gain matrix and the norm of the close-loop transfer function?
- (g) **(PTS: 0-2)** Use output feedback setup to compute the \mathcal{H}_2 and \mathcal{H}_∞ controller replicating the result of LQR controller in Part (a). What are the gain matrix and the norm of the close-loop transfer function?
- (h) **(PTS: 0-2)** Use output feedback setup to compute the \mathcal{H}_2 and \mathcal{H}_∞ controllers minimize x_3 and u with weighting $W = 10^5$ on x_3 . What are the gain matrix and the norm of the close-loop transfer function?
- (i) **(PTS: 0-2)** Compare the transfer function response (Bode) of LQR, \mathcal{H}_2 , and \mathcal{H}_∞ from disturbance to x_3 . 5 Bode plots.
- (j) **(PTS: 0-2)** Simulate the closed loop system of LQR, \mathcal{H}_2 , and \mathcal{H}_∞ for 10s with a 0.1-m step disturbance input starting at 1s with zero initial condition and plot x_3 . 5 simulations.